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Geometric Hysteresis of Alveolated Ductal Architecture

Low Reynolds number airflow in the pulmonary acinus and aerosol particle kinetics therein are significantly conditioned by the nature of the tidal motion of alveolar duct geometry. At least two components of the ductal structure are known to exhibit stress-strain hysteresis: smooth muscle within the alveolar entrance rings, and surfactant at the airtissue interface. We hypothesize that the geometric hysteresis of the alveolar duct is largely determined by the interaction of the amount of smooth muscle and connective tissue in ductal rings, septal tissue properties, and surface tension-surface area characteristics of surfactant. To test this hypothesis, we have extended the well-known structural model of the alveolar duct by Wilson and Bachofen (1982, "A Model for Mechanical Structure of the Alveolar Duct," J. Appl. Physiol. 52(4), pp. 1064–1070) by adding realistic elastic and hysteretic properties of (1) the alveolar entrance ring, (2) septal tissue, and (3) surfactant. With realistic values for tissue and surface properties, we conclude that: (1) there is a significant, and underappreciated, amount of geometric hysteresis in alveolar ductal architecture; and (2) the contribution of smooth muscle and surfactant to geometric hysteresis are of opposite senses, tending toward cancellation. Quantitatively, the geometric hysteresis found experimentally by Miki et al. (1993, "Geometric Hysteresis in Pulmonary Surface-to-Volume Ratio during Tidal Breathing," J. Appl. Physiol. 75(4), pp. 1630–1636) is consistent with little or no smooth muscle tone in anesthetized rabbits in control conditions, and with substantial smooth muscle activation following methacholine challenge. The observed local hysteretic boundary motion of the acinar duct would result in irreversible acinar flow fields, which might be important mechanistic contributors to aerosol mixing and deposition deep in the lung. [DOI: 10.1115/1.4005380]

Keywords: lung, surfactant, smooth muscle, connective tissue, geometric hysteresis, acinar kinematic irreversibility, finite element model, micromechanics

Introduction

The airflow patterns of low Reynolds number acinar gas flow and the fate of inhaled particles suspended therein are largely determined by acinar wall motion. Due to the characteristic structure of the acinar duct, namely a thoroughfare channel surrounded by numerous dead-end air pockets, a central channel flow passing by alveoli induces a slowly rotating flow in a rhythmically expanding/contracting alveolus. The alveolar flow with recirculation can be chaotic under the effect of small disturbances [1]. Perturbation includes (1) the effects of small nonzero Reynolds numbers (inertial effects) studied in, e.g., Tsuda et al. [2] and Henry et al. [3,4], and (2) the effects of a small asynchrony between ductal flow oscillation and cyclic alveolar wall motion studied in, e.g., Haber et al. [5,6] and Haber and Tsuda [7]. The focus of this study is on the latter case.

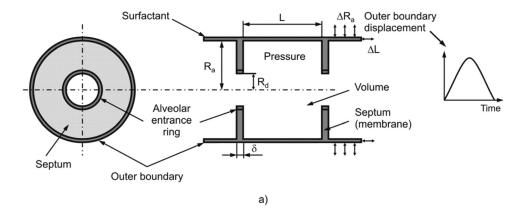
Whereas the principal mode of lung expansion approximately satisfies geometric similarity [8–13], it is known that during tidal ventilation geometric hysteresis does exist, albeit small in magnitude [11]. We have both experimentally and theoretically demonstrated that the presence of such wall motion irreversibility (i.e.

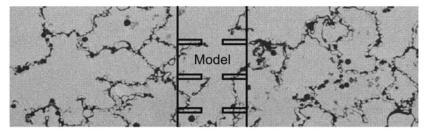
temporal asynchrony), even if small in magnitude, can cause significant acinar flow kinematic irreversibility and consequently enhance aerosol mixing and deposition in the pulmonary acinus [1–7,14–18]. The motivation for this study, therefore, was to identify and quantify the mechanical origins of acinar geometric hysteresis.

Wilson and Bachofen's [19] seminal work showed that acinar airway architecture is maintained by a balance between forces pulling the alveolar duct radially inward (hoop stresses associated with tension borne in the connective tissue and smooth muscle in the alveolar entrance ring), and forces retracting the duct radially outward (primarily associated with alveolar septal surface tension at the air-liquid interface, but which may also include septal tissue tension). Due to cyclic expansion and contraction, these forces behave hysteretically, (e.g., [20,21]), the resulting acinar geometry will display geometric hysteresis depending upon the balance of stress hysteresis of these mechanical elements in series. For example, if all elements mechanically in series display the same time dependence of their stress relaxation or adaptation, then in spite of even large stress hysteresis, there will be no geometric hysteresis. By contrast, to the extent that the time courses of stress adaptation between the ductal connective tissue, surface tension, and septal tissue tension are not strictly proportional, then there will be geometric hysteresis, in the sense of a different fractionation of acinar volume between duct and alveoli on inspiration and expiration, at the same overall lung volume.

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b) Morphology of alveolar space in lung (Gil et al. JAP, 1979)

Fig. 1 (a) Schematic of the model of alveolar duct: left panel – cross-section in the plane orthogonal to the duct longitudinal axis, right panel – axial cross-section of duct; (b) representative geometric location of the model in the acinus, with morphology of alveolar space in lung [10]

We quantified these ideas in the context of a simple geometric model of a rhythmically expanding/contracting alveolated duct, whose alveolar septa are in mechanical equilibrium as described above. We extended Wilson and Bachofen's [19] analytical model, using finite element computational techniques, to incorporate realistic, experimentally based, nonlinearly elastic and hysteretic behaviors of all stress bearing elements.

The analyses showed several patterns of geometric hysteresis (including loop sense reversals as found by Miki et al. [11]), depending on the magnitude of hysteretic characteristics of force bearing elements. In particular, smooth muscle tone in the alveolar entrance ring plays an important role in the geometrically hysteretic behavior of acinar architecture.

Materials and Methods

Model Geometry. Our simplified geometric model of acinar architecture consists of three components (Fig. 1(a)). (1) The outer boundary of the acinus is represented by a long expandable closed-end cylindrical pipe whose kinematic motion is prescribed. The outer boundary corresponds to what is described as primary alveolar septa in development, here referred to as outer wall septa. (2) The septa connecting the outer wall with the alveolar duct (secondary septa developmentally) are represented by annular disks, here referred to as simply radial septa. (3) The duct proper is represented by stress bearing rings at the inner openings. In Fig. 1(b), the manner in which this model represents actual parenchymal architecture is shown by the overlay of the model on a photomicrograph showing the morphology of the alveolar space. The model is axisymmetric, composed of repeated unit cells with outer (acinar) and inner (ductal) radii of R_a and R_d , respectively, and axial length L. Thus, the volume of the unit cell, bounded by the outer wall and two neighboring radial septa (extended to two neighboring disks), is $\pi R_a^2 L$; the surface S of the radial septa is $2\pi(R_a^2-R_d^2)$ (the factor of 2 arises from two radial septa facing the interior of each unit cell). The thickness of the radial septa is denoted by δ .

Material Models. The material properties of each model component are as follows. First, the major material constituents of the alveolar entrance ring are connective tissue and smooth muscle [22-24]. The connective tissue exhibit a nonlinear constitutive law, and is hysteretic with a weak power law dependence on frequency [25,26]. However, at breathing frequencies, its hysteresis is low, which we neglect in comparison with that of the smooth muscle. In short, we take the connective tissue to be essentially elastic (i.e. history independent). By contrast, the smooth muscle response is both nonlinear and displays significant hysteresis [21,27,28]. A typical uniaxial stress/stretch curve for entrance ring connective tissue of cat lung parenchyma is shown in Fig. 2(a) [29]. Connective tissue within the radial and outer wall septa is described by representative uniaxial and biaxial stress/ stretch curves in Fig. 2(b) [29,30]; this is described in more detail below. Stress/stretch behavior of smooth muscle is shown in Fig. 2(c), taken from Sasaki and Hoppin [21]; note the hysteresis is present. The fractionation of the stress born in parallel within the ductal ring between muscle and connective tissue is quantified by a fraction m (approximated by the ductal ring volume density of muscle), where 0 < m < 1. This is a functional fractionation, related to both the morphometric volume fraction and to the degree of activation of the smooth muscle.

The radial septum is modeled as a membrane made of a nonlinear elastic material with hardening characteristics. The constitutive law for the mechanical behavior of homogeneous isotropic membranes can be deduced from uniaxial or biaxial loading experiments (e.g., [30,31]). First, we adopted the uniaxial curve of Fukaya et al. [29] (Figs. 2(a) and 2(b) (right curve)). Because the biaxial curve for alveolar septa has never been measured, we estimated it from the material characteristics of a similar biological

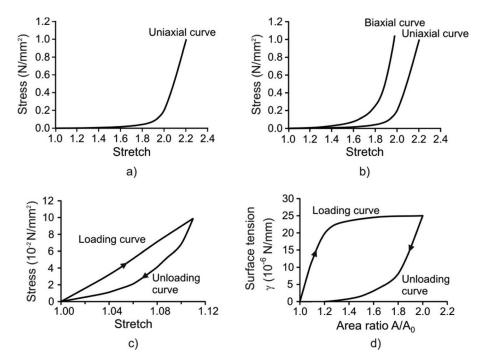


Fig. 2 Material models used in the analysis. (a) Uniaxial model for connective tissue of alveolar ring [29]; (b) Biaxial model for septum tissue [29,30]; (c) Hysteretic model of muscle constituent of ring material [21]; (d) Hysteretic characteristic of surfactant [35].

membrane, cat mesentery, obtained by Hildebrandt et al. [30]. Assuming that the ratio between the biaxial and uniaxial stresses at a given stretch is similar in both tissues, we estimated the biaxial curve of alveolar septa (Fig. 2(b) (left curve)) based on the uniaxial curve measured by Fukaya et al. [29]. For general loading, the stress/stretch states lie between the biaxial and uniaxial curves. Our algorithm (see Appendix) computes stresses for in-plane stretches in the principal strain directions; in our case these are circumferential (or hoop) and radial directions.

The surface of the model is covered by surfactant (Fig. 1(a)). The air-liquid interface generates surface tension γ , which is an external load on the tissue surface. We assume a no-slip condition between the surfactant layer and the underlying tissue; this implies that the surface tension may vary locally as it follows potentially non-uniform strain in the radial septa. The surface tension-surface area relationship exhibits hysteretic behavior during cyclic motion. It has been measured surfactometrically from bronchoalveolar lavage (BAL) fluid extracts [20], by models and inferences from pressure-volume (PV) curves and morphometry in excised and intact lungs [32-35], and by direct observation of the spread of liquid droplets [33,36]. The constitutive relationship (Fig. 2(d)), calculated by Wilson [35] based on PV curves and morphometry in rabbits [10,37] is used in this study. Note that all material models shown in Fig. 2 rely on data obtained under quasi-static conditions. The above is a simplification; indeed, the hysteretic behavior of surfactant during quiet breathing remains controversial [38,39]. Nevertheless, to the extent that we choose to lump all the septal hysteresis in the surfactant layer, given that it is mechanically in parallel with septal tissue with a no-slip condition, this artificial compartmentalization is likely to give rise to only minor differences in behavior.

Boundary Conditions. The outer wall septa of the model expand and contract in a geometrically similar fashion, representing the principal mode of lung expansion [8–13]. We take the time dependence to be sinusoidal, although the temporal evolution in this work is entirely a sequence of quasi-steady states; there are no dynamics per se, and time plays only a role of a parameter

which labels the volume. We write the radial and axial motions of the outer acinar boundary, $R_a(t)$ and L(t), as

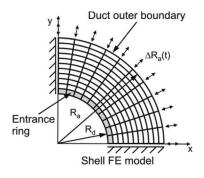
$$R_{\rm a}(t) = R_{\rm mean} - (1/2)\Delta R_{\rm a}\cos t$$

$$L(t) = L(0)R_{\rm a}(t)/R_{\rm a}(0)$$
(1)

where R_{mean} and ΔR_a are the mean radius and the peak-to-peak amplitude of the outer boundary radial displacement, which we take numerically to be 328 µm and 176 µm respectively. Note that the peak-to-peak amplitude of radial displacement corresponds to the volume change from minimal volume to total lung capacity, similar to simulations by Denny and Schroter [40]. We take the aspect ratio of alveoli to approximate spherical symmetry, and set $L(0) = R_a(0) - R_d(0)$. The initial radius of the duct $R_d(0)$ is taken as 120 μm . The initial thicknesses of both the radial and outer wall septa are taken to be $5 \mu m$, and the initial cross-sectional area of the alveolar entrance ring to be $25 \,\mu m^2$ [22,32]. As noted above, time is only a parameter in this model, for convenience we take the period as 2π . Minimal volume occurs at time zero, at which point all elements (including surfactant) of the microstructure are assumed to be stress free. This corresponds to the zero transpulmonary state in an excised preparation.

Note that we are taking displacements as boundary conditions. From the whole organ point of view, lung displacements are driven by variations in pleural pressure, and therefore pressure boundary conditions would be most appropriate. On the other hand, at this level of modeling the geometric characteristics of the acinus, and in particular its geometric hysteresis, there is necessarily a unique relationship, albeit history dependent, between volume displacements and pressure. In that sense, dealing with volume displacements as boundary conditions as in Eq. (1) and computing the resulting pressures, is equivalent to setting pressure variations as the independent variable and computing the resulting volume displacements.

The Governing Equations for the Radial Septa. The quasistatic deformations of the radial septum and entrance ring can be determined by solving the two equilibrium equations for stresses



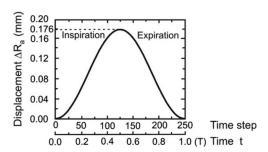


Fig. 3 Finite element model of alveolar duct. Septum is modeled by membrane (shell) finite elements and ring is modeled by line finite elements. Deformations of ring and septum are calculated for prescribed radial displacement of the outer (acinar) boundary $\Delta R_a = \Delta R_{a,max} \sin t$. Time is a parameter controlling volume history; no dynamics are associated with it. Solution is obtained incrementally using 250 steps over the breathing cycle.

 σ_{xx} , σ_{yy} , σ_{xy} , in local Cartesian coordinates x and y in the septal plane:

$$\frac{\partial \sigma_{xx}}{\partial x} + \frac{\partial \sigma_{xy}}{\partial y} = 0$$

$$\frac{\partial \sigma_{xy}}{\partial x} + \frac{\partial \sigma_{yy}}{\partial y} = 0$$
(2)

The task in our analysis is to calculate, for a given radial displacement and history of the outer wall (the independent variable), the radial displacements of the alveolar entrance ring and septum in a way that satisfy these equilibrium equations together with the stresses and strains within the tissue and surfactant obeying the (hysteretic) constitutive laws.

Numerical Model. The magnitude of strains that we simulate is too large for use of a linear approximation schemes. We use a displacement based finite element numerical method [41-43] and implement the features of nonlinear, hysteretic materials into the finite element program PAK [44]. One quarter of the septum is considered with imposed symmetric boundary conditions along the x and y axes (Fig. 3). Note that by virtue of circumferential symmetry of the radial boundary conditions at R_d and R_a , the solution inherits this same symmetry. We specify the radial

displacement at the outer septum radius as shown in Fig. 3 as arrows indicating displacements of R_a at the boundary.

Briefly, the alveolar entrance ring is modeled by line finite elements connecting the finite element nodes on the internal rim. In calculating the nodal forces resulting from the alveolar ring deformation, we use the material characteristics of each constituent, shown in Fig. 2(c) for muscle and in Fig. 2(a) for connective tissue. The alveolar septal membrane is modeled by isoparametric shell/membrane finite elements, with a biaxial material model (Fig. 2(b)). Surfactant is modeled by calculation of stresses in tissue due to surfactant, and then the nodal forces of the line and shell finite elements generated by surface tension (Fig. 2(d)) are evaluated. The stresses in tissue at a given point, caused by surface tension, depend on the history dependent local surfactant area strain of the surfactant surface at that point (see Appendix for details)

Our model consisted of 70 membrane elements and 10 line elements for the entrance ring with 88 nodal points. The number of steps in the incremental analysis was 250 (125 for inspiration and 125 for the expiration regime). Implicit incremental procedure [41,43] and implicit stress integration scheme of the constitutive relations [42,45] where implemented as the methods which provide the best solution accuracy; details are given in the Appendix. We used tight equilibrium iteration tolerance of 10^{-6} in order to secure solution accuracy at each load step.

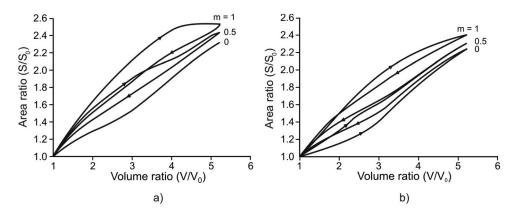
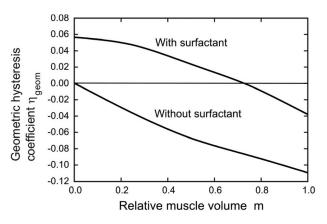


Fig. 4 Geometric hysteresis within the alveolar duct. (a) Case without surfactant. Geometric hystersesis is due to hysteresis of muscle constituent within the alveolar entrance ring and it increases with increasing m. The S-V hysteresis loop has a clockwise direction for one breathing cycle. (b) Case with surfactant. The S-V hysteresis loop due to surfactant has counterclockwise direction (condition with no muscle constituent, m=0). Since S-V hysteresis loops corresponding to muscle and surfactant have the opposite senses, the resulting hysteresis is smaller when m=0.5 with respect to m=0 and the loop changes the direction for m=1.



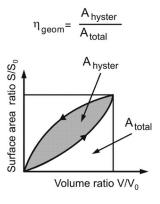


Fig. 5 Coefficient of geometric hysteresis η_{geom} plotted against m; its geometric construction is shown in the cartoon. In the absence of muscle contribution, the hysteresis loop is entirely controlled by surfactant, and the loop is counterclockwise with positive η_{geom} . Its value progressively decreases with muscle activation, crossing zero into a regime dominated by muscle, where the loops are clockwise and η_{geom} is negative. Also shown is the hysteretic behavior in the absence of surfactant, showing the uniformly negative η_{geom} .

Quantifying Geometric Hysteresis. We define a coefficient of geometric hysteresis η_{geom} by the ratio of the loop area on the surface area-volume plot to the area of the bounding rectangle enclosing the loop. This definition is an exact analogy to the shape factor for pressure-volume hysteresis introduced by Bachofen and Hildebrandt [46] and extended to non-elliptical curves by Fredberg and Stamenovic [47] (this construction is shown graphically as a cartoon in Fig. 5). Trajectories of material points on the internal architecture display hysteretic character over breathing cycles (Fig. 7) and we have introduced a measure of hysteresis in displacements, the displacement hysteresis coefficient S_{hyster} (Fig. 8), in an analogous way as for the area-volume hysteresis.

Results

Geometric hysteresis represented by a septal surface - duct volume relationship is shown in Fig. 4. When surfactant is absent (Fig. 4(a)), the degree of geometric hysteresis in the S-V loop depends entirely on the amount of muscle constituent in the alveolar entrance ring; no S-V geometric hysteresis occurs with m = 0. As the amount of muscle fraction increases, the magnitude of geometric hysteresis increases (see m = 0.5, 1 in Fig. 4(a)). Note that the S-V hysteresis associated with muscle is clockwise in the S-V loop. By contrast, the S-V hysteresis associated with surfactant (in the absence of muscle, m = 0) is counterclockwise in the S-V loop (see m=0 in Fig. 4(b)). A mechanism responsible for these opposing loop directions is described in the Discussion section. With surfactant, the extent of geometric hysteresis decreases as the amount of muscle increases (e.g., m = 0.5), but is still dominated by the hysteretic characteristics of surfactant seen because the overall S-V loop remains counterclockwise (see m = 0.5 in Fig. 4(b)). With further increase in the amount of muscle (the limit m=1), however, the S-V loop changes its direction to largely clockwise, showing a dominant role of muscle-driven hysteresis over surfactant-driven hysteresis (see m = 1 in Fig. 4(b)). Finally, it should be noted that these directional changes of geometric hysteresis due to the relative contribution of muscle-driven hysteresis to surfactant-driven hysteresis have indeed been observed in the lungs of live rabbits by Miki et al. [11] (see the Discussion section).

In Fig. 5, the coefficient of geometric hysteresis η_{geom} is plotted with respect to muscle fraction in alveolar duct, m, for cases with and without surfactant. Note that our sign convention is to take A_{hyster} and hence η_{geom} positive or negative as the S-V loop is counterclockwise or clockwise, respectively. Without surfactant, η_{geom} is negative, and quantifies the clockwise S-V hysteresis

entirely due to the hysteretic characteristics of muscle in the alveolar entrance ring. With surfactant present, on the other hand, η_{geom} is positive for lower m values, consistent with the counterclockwise S-V hysteresis being dominated by surfactant; however, η_{geom} becomes negative for higher m values, showing a progressive domination by muscle. These two effects are balanced at a muscle fraction of $m \approx 0.72$, at which no net geometric hysteresis is observed.

Geometric hysteresis observed in the *S-V* loop (Figs. 4 and 5) are indeed originated from the hysteresis (or asynchrony) in radial septal displacements. This will be shown next in detail. We present change of the septal radii over one cycle, trajectories of material points of the septum, and radial displacement hysteresis (Figs. 6–8, respectively). The results are obtained assuming that the septum is covered by surfactant, and that the alveolar entrance ring consists of two extreme cases of the muscle content: m = 0 (no muscle) and m = 1 (purely muscle).

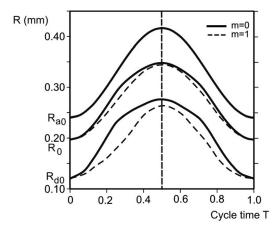


Fig. 6 Radial displacements of internal rim (radius R_d), outer boundary (radius R_a), and points at initial radius $R_0=190~\mu m$ during one cycle; for two extreme cases: condition with no muscle constituent (m=0), and with only muscle constituent within the ring (m=1). Due to hysteretic characteristic of surfactant, and muscle radial displacement curves are not symmetric with respect to the middle line (end of inspiration), except for the outer boundary, with asymmetry more pronounced for the domain closer to internal rim. Displacements and asymmetry are smaller for m=1.

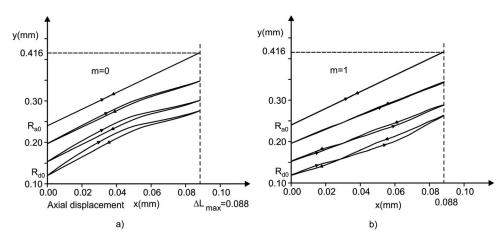


Fig. 7 Particle trajectories within radial plane (x-y) of material points at outer boundary, internal rim and at two membrane radii. Axial maximum displacement is $(\Delta L)_{\max} = 0.5$ $(\Delta R_d)_{\max}$ following from geometrical similarity during alveolar deformation (see equation 1). (a) Case with no muscle constituent (m=0): the trajectories display clockwise hysteretic loops due to dominant surfactant action during inspiration (see Fig. 2(d)), with loops diminishing from the internal to the external membrane boundary; (b) Case m=1: The muscle hysteresis acts in the opposite sense from the surfactant effects (see Fig. 2(c)), and hysteretic loops become counterclockwise in the regime where muscle hysteretic effect is dominant (expiration regime); the loops diminish when approaching to external boundary.

Considering hysteresis in displacements, we first show radial displacements of the alveolar entrance ring (radius R_d), the model outer boundary (radius R_a), and a point close to the middle radius (initial radius $R_0 = 190 \,\mu\text{m}$) during a breathing cycle, as shown in Fig. 6. While the time course of the outer boundary (R_a) is exactly symmetric with respect to the middle line (T=0.5), which is dividing inspiration-expiration regimes, representing the imposed condition (see Eq. (1) and Fig. 3), the time course of other material points at any radial position R ($R_a \le R < R_d$) on the septum exhibits asymmetry. The maximum asymmetry occurs at the inertial rim (R_d) and the extent of asymmetry monotonically decreases as R approaches R_a . The cases of no muscle content at the entrance ring, m = 0, are shown by solid lines, while the cases of m=1 (totally muscle) are represented by dashed lines. A comparison between two cases shows that displacements and asymmetry are smaller for m = 1, indicating that hysteresis driven by smooth muscle acts in the opposite sense with respect to the surfactant hysteretic action.

Trajectories of septal material points in a radial x-y plane are shown in Fig. 7. Radial displacements of selected points (shown in the y-direction in the figure) are obtained from the FE model, while axial displacements (in the x-direction in the figure) are calculated using Eq. (1). Trajectories in Fig. 7(a) (no muscle content, m=0) have hysteretic character of surfactant (except for the outer boundary with prescribed displacements), with the clockwise hysteretic loop. This clockwise loop characteristic is in agreement with the counterclockwise S-V hysteresis in Fig. 4(a) since smaller septal surface corresponds to larger radial displacements. Similarly to the phenomenon shown in Fig. 6, the hysteresis becomes smaller when the material points approach the outer boundary.

In Fig. 7(b) the trajectories of septal material points are shown, assuming m = 1. With the presence of both surfactant and muscle, the hysteretic effects of surfactant and muscle are superimposed. We have hysteretic trajectories with loops which change sign over the cycle. The loop of the inner rim is clockwise at the start of inspiration since the surfactant effects are dominant; contrary,

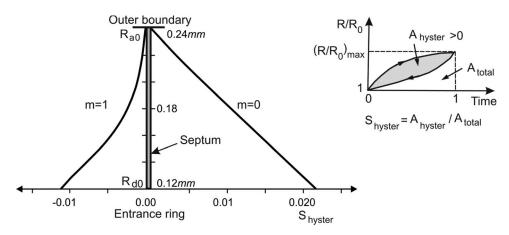


Fig. 8 Distribution of displacement hysteresis coefficient S_{hyster} . The ratio R(t)/R(0) displays a hysteretic character over a breathing cycle, and a measure of this hysteresis is defined as the ratio $S_{hyster} = A_{hyster}/A_{total}$. (a) Case without muscle content within the ring (m=0): the hysteretic loops are clockwise (considered positive) since displacements are larger during inspiration, and diminish toward the external boundary; (b) Case m=1: the overall hysteretic muscle characteristic is dominant and the surface A_{hyster} is negative (see Fig. 7(b)).

toward the end of expiration, the muscle effect dominates and the loop becomes counterclockwise. As in case m = 0, the loops become smaller when approaching the duct boundary because the harmonic radial displacements are prescribed at the outer boundary.

An insight into a distribution of hysteresis in displacements along the septum radius can be seen from Fig. 8. In analogy to the definition of the coefficient of geometric hysteresis η_{geom} (see Fig. 5), we introduce the displacement hysteresis coefficient S_{hyster} as the ratio of the loop area on the displacement-time curve to the area of the bounding rectangle enclosing the loop. Here, we have plotted values of the S_{hyster} over the septum radius. In case of hysteresis due to surfactant only (with no muscle content (m = 0)), Shyster approximately linearly increases from no hysteresis $(S_{hyster} = 0)$ at the external boundary to the maximum hysteresis at the internal radius; S_{hyster} is positive showing that hysteresis loops are clockwise (graph on the right). Conversely, when with m = 1, S_{hyster} is negative (graph on the left), indicating that the hysteresis due to the muscle dominates over to that due to the surfactant, S_{hyster} is negative and nonlinearly increases (in value) from $S_{hyster} = 0$ at the outer boundary to the internal rim.

Discussion

The principal findings of this study are as follows: (1) the alveolar architecture exhibits geometric hysteresis due to the stress hysteretic characteristics of smooth muscle and surfactant. The hystereses of smooth muscle and surfactant act in opposing senses in terms of inducing geometric hysteresis, and tend toward quantitative cancellation; (2) the hysteretic characteristics of smooth muscle and/or surfactant produce asynchrony and hysteresis in radial septum displacements.

Methodology - A Brief Overview of Past Models and Our Basic Computational Approach and Model Limitation. In the past, a number of acinar microstructural models have been reported. Traditionally, it was first considered that the major load-carrying structural elements were the alveolar walls and the alveolar air-liquid interface covering the septal wall membranes. These types of "membrane" models with various configurations of space-filling polyhedrons were widely discussed (see a review by Fung [48]). In the early 1980 s, the importance of the cables of fibers weaving through the alveolar ducts, especially through the alveolar entrance rings, was stressed as the other major load-carrying elements maintaining the alveolar microstructure [19]. Since then, several "line" (or "line-membrane") models have been proposed (e.g., [40,49-55]). Among them, there are a few analytical models reported (exceptions include Budiansky and Kimmel [49]; Kimmel and Budiansky [54]), chiefly due to analytic limitations associated with the nonlinearity of material characteristics. An alternative computational approach, especially in terms of incorporating experimentally obtained material properties (through constitutive laws) of microstructural elements into the structural model, has become more popular. For instance, one of the most complete computational models currently available in the literature includes the finite element model of Denny and Schroter ([40,53]). The model consists of 36 truncated octahedra, with line elements (composed of elastin and collagen) located at septal borders and within the septal walls. Surfactant with hysteretic characteristics is also included, as well as geometric hysteresis (a main difference between these and our model, besides that our model includes smooth muscle with hysteretic characteristics in the alveolar entrance ring, is discussed below).

In the present study, the fundamental model design is based on the idea of Wilson and Bachofen [19] that there are competing force-bearing structures in acinar microstructural architecture. Four dominant force bearing elements of acinar architecture were stretched into their nonlinear regime and the quasi-static motion of the model alveolar microstructure was solved numerically using state-of-the-art custom built finite element (FE) computational techniques (see, e.g., Refs. [43,45]). In particular, our model, unlike others (including the model by Denny and Schroter [40,52,53]), is designed to probe the following specific characteristic: since a force balance between the alveolar entrance ring inward force and alveolar septal outward surface force is demanded by equilibrium and because the hysteretic constitutive laws of the ductal tissue and radial septa (including surfactant) are different, the alveolar entrance ring and septum expand and contract asynchronously with respect to the motion of the ductal outer boundary. The consequence of this asynchrony is that the acinar architecture at the microstructural level does not deform in a geometrically similar fashion to the boundary deformation. This is likely to have a significant effects on aerosol mixing and deposition in the pulmonary acinus [1–7,14–18].

The model limitations rely on several simplifications, which might be classified into two groups: geometrical and material. It is assumed that the duct is an ideal circular cylinder, with the septum and internal ring also being axisymmetric and circular. It is taken that the material of septum and ring is isotropic with hardening and hysteretic characteristics, described by idealized constitutive curves; representation of surfactant mechanical response is also taken using idealized hysteretic constitutive curves.

Geometric Hysteresis. The principal contribution of our model is in the quantitative fractionation of geometric hysteresis among the stress hysteretic behavior of smooth muscle and surfactant.

Mechanical response of the alveolar duct internal architecture during duct expansion and relaxation cycles is determined as a result of the superposition effects between: hardening character – nonlinear increase of stiffness with deformation of the septal tissue (Fig. 2(b)); hardening and hysteretic characteristic of the ring two-component material (Figs. 2(a) and 2(c)), and hysteretic character of the surfactant (Fig. 2(d)).

During breathing cycles, the ducts harmonically expand and extend and then come back to the initial (undeformed) configuration, producing deformation of internal architecture. The membrane and ring, used in our model as the structural elements of the duct interior, deform over cycles. Surfactant, which covers the tissue, changes its surface area (as the septal area changes) and produces a distributed loading on the tissue due to surface tension at the liquid-tissue interface. The surface tension has a hysteretic character and therefore the loading within a cycle is hysteretic: the surface tension is higher during inspiration than during expiration (Fig. 2(d)), thus it tends to produce larger deformation. As such, it results in larger radial displacements during inspiration. Therefore, an amount of surface area change due to surfactant force tends to be more pronounced in inspiration than in expiration.

On the other hand, the muscle is tensed during inflation and relaxing during deflation. The stresses within the alveolar entrance ring are larger during inspiration (Fig. 2(c)) so that the alveolar ring muscle produces larger resistance for the septum to deform, and therefore its hysteresis acts in the opposite sense with respect to the surfactant hysteresis. The resulting hysteretic response of the internal duct elements depends on which hysteretic action is dominant at a certain moment within the breathing cycle.

Septal Surface -Duct Volume (*S-V*) **Hysteresis.** The observation that hysteresis induced by smooth muscle is manifested as a clockwise *S-V* loop (Fig. 4(*a*)) is consistent with the fact that muscle tension is increased during inflation compared with expiration. The structural stiffness is larger during radial expansion of the duct during inflation, promoting an increased area of the radial septa relative to that at the same volume during expiration. Thus, the net effect results in a clockwise rotation in the *S-V* loop. By contrast, surface tension hysteresis has the opposite effect. Septal surface tension at the air-liquid interface acts to retract the duct

radially outward. Therefore, the sharply increased surface tension of surfactant during its areal expansion phase of inspiration compared with the low surface tension during areal compression in expiration implies an increased septal radial expansion associated with decreased area during inspiration compared with expiration. This results in a counterclockwise rotation in the *S-V* loop.

We express the overall geometric hysteresis by the index η_{geom} , given by the ratio of (signed) geometric loop area to the bounding rectangle area of the surface area - volume relationship (Fig. 5, cartoon). Under the physiologically realistic conditions when both muscle and surfactant effects are present, the S-V loop exhibits a combination of the two extreme cases described above, sometimes resulting in a complex pattern, such as a figure "8," which appears in Fig. 4(b), depending on the different histories/combinations of muscle and surfactant contributions. Interestingly, the S-V hysteresis due to these competing elements are counterbalanced at a muscle fraction of approximately 0.72 resulting in no overall geometric hysteresis ($\eta_{geom} = 0$). In addition to the global index of hysteresis given by η_{geom} (through the area history), we note that our technique here also gives unambiguous predictions about the history of local material points as well, in particular, the irreversible trajectories of the boundary surfaces of the alveolar duct (Fig. 7).

Finally, it is important to note that the complex interplay between the two hysteretic components described above was observed in experiments. Miki et al. [11] found that geometric hysteresis does exist, albeit small in magnitude, in live adult rabbits. Quantitatively, they found that, at isovolume points during tidal breathing, the fractional change in optical mean free path for diffuse light scattering was roughly -0.16 (using the opposite sign convention from here). Using the approximate inverse quadratic dependence of mean free path on surface to volume ratio, and using the relationship between hysteresis measured as fractional differences at isovolume points and the loop area ratios, their results translate into a geometric hysteresis index of $\eta_{geom} \approx +0.06$. This value is strikingly close to the limiting value that we compute for essentially complete domination of geometric hysteresis by surface tension behavior of surfactant, with little role for smooth muscle. By contrast, Miki et al. [11] also found loop reversals when the animals were challenged with the contractile agonist methacholine. In that circumstance, their results translate to a geometric hysteresis index of $\eta_{geom} \approx -0.05$, indicating that in this case there is a substantial contribution from activated smooth muscle, of a comparable but larger magnitude (the loop is reversed) compared with the contribution from surfactant.

The Role of Smooth Muscle at the Alveolar Level. The role of smooth muscle in lung mechanics has been extensively studied at the level of airways in the bronchial tree, especially in the context of asthma. Less is known, however, about the quantification and role of smooth muscle in the alveolated duct (please note Refs. [12,24,56]. Statically, it is believed to play a role in maintaining alveolar microarchitecture (e.g., Smith and Stamenovic [34], Stamenovic and Smith [57]), but its contribution to geometric hysteresis in compensating for that associated with surfactant has not been probed. Here we have made a first attempt to address this question. On the other hand, while we are confident of the extremes, we have computed for the displacements and surface-volume relationship and we note that serious caution must be exercised with respect to the quantification in terms of muscle fraction m (note that the value of mappears to be species dependent; for instance, guinea pigs apparently have more muscle than other small animals [24].) Not only is the morphometric distribution of m given by a limited database, but it is largely unknown how to translate such morphometric measurements directly into stress bearing implications. We thus suggest interpreting numerical values of m as a functional index of the contribution to mechanical equilibrium in the duct.

Hysteresis in Radial Displacements. The asymmetry in radial displacements with respect to inspiration versus expiration (Fig. 6), and the hysteretic trajectories (Fig. 7) found in this study result in asynchronous motion of the surfaces of the acinar structure. This has an important consequence. Since the surfaces of the acinar structural element represent the boundaries of the air domain in the alveolar microstructure, its motion is directly coupled to air flow. In particular, its asynchronous motion induces aerosol mixing and deposition deep in the lung.

These asynchronous motions of the acinar structure are results of the effects of hystereses of surfactant and muscle. We present results for the two extreme conditions corresponding to m=0 (no muscle constituent within the alveolar entrance ring) and m=1 (only muscle constituent, with no connective tissue); all other cases will fall between these two extremes. It is interesting to note that the hysteretic effects of the surfactant and muscle content tend to cancellation.

The Effects of Geometric Hysteresis on Acinar Flow. The effects of geometric hysteresis on acinar fluid mechanics, consequently aerosol mixing and deposition are substantial, such a phenomenon has been demonstrated previously [1–7,14–18]. In particular, the presence of geometric hysteresis is crucial in the case when an alveolar flow exhibits recirculation because it can break alveolar flow into chaos (e.g., [1,2]). In this regard, the occurrence, rather than its extent, of geometric hysteresis is important as a source of perturbation [1]. The detailed discussion regarding the onset of chaos in the pulmonary acinus is beyond the scope of this paper; such a discussion can be found elsewhere (Tsuda et al. [1,18]).

Summary

In this study, we probed the origins of geometric hysteresis at the level of the pulmonary acinus. We used realistic experimental relationships for the constitutive laws of parenchymal tissue, smooth muscle, and surfactant in a Wilson-Bachofen type model of the alveolar duct. With finite element techniques, we solved for the resulting behavior during vital capacity maneuvers, focusing especially on the displacement hysteresis and the surface area volume history dependence. We found striking agreement between our results in the area - volume hysteresis and the magnitude of geometric hysteresis observed by Miki et al. [11], including hysteretic loop reversal phenomena with smooth muscle stimulation by contractile agonist. These results, including quantified hysteresis in displacements and local irreversibility of the boundary surfaces of the duct, are important not only for elucidating acinar micromechanics but also for forming the foundation for coupling local hysteretic boundary motion with irreversible acinar flow fields, which in turn are important mechanistic contributors to aerosol mixing and deposition deep in the lung.

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Appendix

Finite Element Model of Duct and Computational Procedure.

Basic Equations. The basic steps of the numerical procedure are as follows. By applying the principle of virtual work, the equilibrium equations are derived for a finite element at some current state of deformation, with appropriate linearizations due to geometric and material nonlinearities [41–43]. These equations are combined to obtain the equilibrium equations for the whole FE

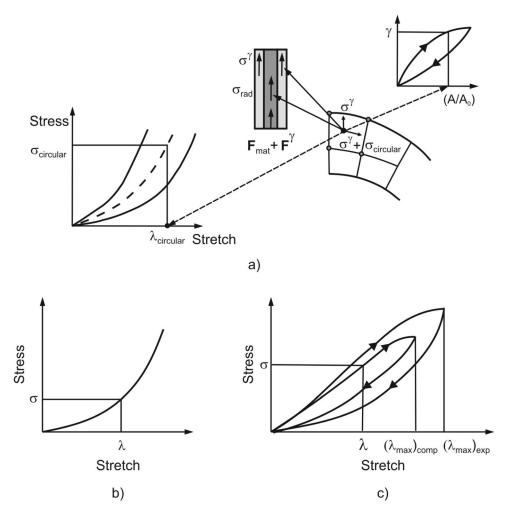


Fig. 9 Sketch of stress calculation. Stresses correspond to the previous iteration (indexed by i-1 in equation (A1)). (a) At each material point of the radial septum the stress is represented as a sum of the stresses due to material deformation (radial stress σ_{rad} and circumferential stress $\sigma_{circular}$) and due to surfactant σ^{7} . These stresses are evaluated from constitutive laws for material (biaxial model) and surfactant (hysteretic model). The stresses produce the finite element nodal forces $F_{int} = F_{mat} + F^{7}$ entering the equilibrium equations for the finite element assemblage. (b) and (c) stresses within the ring evaluated for nonlinear elastic and hysteretic material models.

model. The matrix form of equilibrium equation for a finite element in an incremental-iterative solution procedure is, at fixed volume (indexed by time),

$$\mathbf{K}^{(i-1)}\Delta\mathbf{U}^{(i)} = \mathbf{F}_{ext} - \mathbf{F}_{int}^{(i-1)}$$
(A1)

where $\mathbf{K}^{(i-1)}$ is the element stiffness matrix, $\Delta \mathbf{U}^{(i)}$ is the change in nodal point displacements at the i^{th} iteration, \mathbf{F}_{ext} is the vector of external forces, and $\mathbf{F}_{int}^{(i-1)}$ is the internal element nodal force vector. Note that $\Delta \mathbf{U}^{(i)}$ is determined by the stiffness and internal forces at iteration i-1. Iterations continue until the differences of external and internal forces are smaller than a selected error tolerance (at which the displacement increments become correspondingly small).

The solution accuracy depends on the accuracy of the calculation of internal forces [42,43]. These are given, for each i, by

$$\mathbf{F}_{int}^{(i)} = \int_{V(i)} \mathbf{B}_{L}^{(i)T} \boldsymbol{\sigma}^{(i)} dV \tag{A2}$$

where $\mathbf{B}_L^{(i)}$ is the matrix relating the strain increments and increments of nodal displacements, and T denotes matrix transpose. $\boldsymbol{\sigma}^{(i)}$ is the stress tensor, and the integration is performed over the

FE volume $V^{(i)}$. The stresses are calculated at material points of tissue and surfactant, following the corresponding constitutive law which is represented by a material model (Fig. 2). Values of the stresses at any time step (equivalently, any volume step) depend on the current state of deformation (Fig. 9).

Stress Calculation. The stress calculation at a material point of septum is graphically shown in Fig. 9(a) and corresponds to the previous iteration's configuration. The septum is biaxially stretched during duct deformation, and the circumferential and radial stresses are evaluated using the biaxial material model shown in Fig. 2(b). Besides the stresses due to material deformation we have the stress due to surfactant. This is expressed by surface tension γ (as a force per unit length), but it also can be represented by the stress σ^{γ} (= $2\gamma/\delta$, where δ is the septum thickness) acting at each material point of tissue. It is assumed that the stress tensor σ^{γ} is isotropic, and hence has the same value σ^{γ} in all directions within the surfactant surface. Therefore, the total radial and circumferential stresses used for calculation of nodal forces are represented by sum of stress due to material deformation (σ_{rad} and $\sigma_{circular}$) and the stress σ^{γ} .

For the entrance ring modeling, we follow experimental results which show that the ring consists of connective tissue and smooth muscle, with material characteristics and hysteretic constitutive

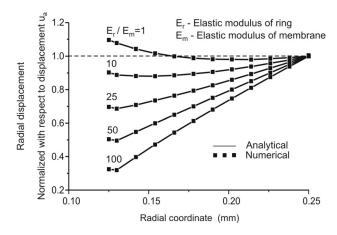


Fig. 10 Validation of methodology: Comparison of analytic and finite element solutions for distribution of radial displacements along the radius

law shown in Figs. 2(a) and 2(c). If the volumetric ratio of muscle, or the relative muscle volume, is $m = V_m/V_{ring}$, where V_m and V_{ring} are the volumes of muscle and ring, then the stress σ_{ring} in the direction normal to the ring cross-section is

$$\sigma_{ring} = m\sigma_{muscle} + (1 - m)\sigma_{tissue}$$
 (A3)

where σ_{muscle} and σ_{tissue} are stresses within the muscle and connective tissue, respectively. For any given stretch of ring, the stresses σ_{muscle} and σ_{tissue} are evaluated from the constitutive laws as schematically shown in Fig. 9(b) and 9(c). The stress σ^{γ} is added to stress σ_{ring} to calculate the nodal forces according to Eq. (A2).

Computational details for stress evaluation for all material models and surfactant are given in Kojic et al. [43,45].

Finite Element Model Verification. Although finite element methodology is well established and confirmed, even for very complex nonlinear problems, for the sake of completeness we give here a comparison of analytical solutions [58] and our numerical solutions obtained with the FE package PAK [44]. Figure 10 shows agreement between analytical and FE solutions for distributions of radial displacements in case of linear elastic material models for ring and septum, for several values of Young's modulus ratios.

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